

SIMPLIFIED NUMERICAL MODEL DEVELOPMENT OF LIGHTWEIGHT-CONCRETE ENCASED COLD-FORMED STEEL ELEMENTS

> By Nathalie Eid Ph.D. candidate

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BUDAPEST UNIVERSITY OF TECHNOLOGY AND ECONOMICS Faculty of Civil Engineering - Since 1782

Department of Structural Engineering

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INTRODUCTION ABOUT COLD-FORMED STEEL SECTIONS



INTRODUCTION





Buckling modes of cold-formed sections (CFS): (a) local, (b) distortional, (c) global and (d) local-flexural interactive modes [Ye et al., 2018]

Innovative building system [Hegyi and Dunai, 2016]





Loading frame of the member tests

A: E= 115.36 MPa

X: E=

270.39 MPa



b) 600 mm; c) 2000 mm

Cross-section of braced elements



Local failure at the end of the braced specimen

specimen type	Number of specimens ^a	Ultimate load [kN]			Increment [%]	
		Unbraced – 0	Braced – A	Braced – X	Braced – A	Braced – X
C90-10-300	3/3/2	27.34	38.93	40.26	42	47
C90-10-600	3/2/3	27.34	37.80	37.44	38	37
C90-10-2000	3/3/3	17.76	34.44	33.57	94	89

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POLYSTYRENE AGGREGATE CONCRETE (PAC)





Expanded Polystyrene beads

Ec= [30-300] MPa

- Poor strength
- Light weight
- Low thermal conductivity

Polystyrene concrete block

RESEARCH PROBLEM

CFS encased in lightweightconcrete are new elements

RESEARCH AIM

- Development of a simplified shell finite element model
- Validation of the model by test results
- Providing rules for the simplified model

Checking by experimental test

Checking by numerical model

New design rules



new structural system

NUMERICAL MODEL DEVELOPMENT FOR UNBRACED SPECIMENS

Experimental and numerical axial loaddisplacement curves of:



the 300 mm column



COMPARISON OF TESTS AND FEA ULTIMATE LOADS

Column	Ultimate load [KN]				
[mm]	Test	FEA	FEA/Test		
300	27.34	27.59	1.01		
600	27.34	27.43	1.00		
2000	17.76	17.80	1.00		



Comparison of local failure mode between test and FEA for 300 mm specimens

SIMPLIFIED NUMERICAL MODEL DEVELOPMENT FOR BRACED SPECIMENS



Applied load and restraint conditions for braced specimens

Parametric study:

- Predict the bracing effect of concrete
- Determine the optimum range of bracing

K = 0.001-100000 N/mm







Failure mode of the 2000 mm column



SPRING STIFFNESS FORMULA DEVELOPMENT FOR PAC BRACED MEMBERS

[Hegedűs and Kollár, 2006]

$$k_{w} = \frac{2\pi E_{c}}{(3-v_{c})(1+v_{c})} \sqrt{\frac{1}{l_{x}^{2}} + \frac{1}{l_{y}^{2}}} \longrightarrow \frac{D}{t} \left(\frac{\partial^{4}w}{\partial x^{4}} + 2\frac{\partial^{4}w}{\partial x^{2}\partial y^{2}} + \frac{\partial^{4}w}{\partial y^{4}}\right) - \sigma_{x}\frac{\partial^{2}w}{\partial x^{2}} - 2\tau_{xy}\frac{\partial^{2}w}{\partial x\partial y} - \sigma_{y}\frac{\partial^{2}w}{\partial y^{2}} = \frac{k_{w}}{t}w$$

$$w = A_{0} \cdot \sin\frac{\pi x}{l_{x}} \cdot \sin\frac{\pi y}{l_{y}}$$

Internal compressed element model [Hegyi, 2016] $\sigma_{cr,p} = 4 \frac{\pi^2 E_s}{12(1 - v_s^2)(b/t)^2} - \frac{4680 MPa}{b/t} + 2.35E_c + \sqrt{E_c.3025 MPa}$

Outstand compressed element model

$$\sigma_{cr,p} = 0.43 \frac{\pi^2 E_s}{12(1 - v_s^2)(b/t)^2} + \frac{1200 MPa}{b/t} + 2.22E_c + \sqrt{E_c.3390 MPa} - 40MPa$$

SPRING STIFFNESS FORMULA DEVELOPMENT FOR PAC BRACED MEMBERS INTERNAL COMPRESSED ELEMENT MODEL

Buckling mode of unbraced (left) and braced (right) internal compressed plates (b/t=50)

Parameter range of numerical investigation

Parameter	Range		
Plate slenderness (b/t)	Internal	Outstand	
	50-250	20-60	
Elastic modulus of PAC (E _c)	50-200 [MPa]		
Element size (A)	5-20 [mm]		



Applied load and boundary conditions for simply supported unbraced and braced plates







OUTSTAND COMPRESSED ELEMENT MODEL



The 3D curve of K as a function of Ec and A (b/t=50)

b/t	a $\left[\frac{N}{mm}\right]$	$b\left[\frac{N}{mm^3}\right]$	C [mm]	d $\left[\frac{N}{mm^5}\right]$	$e\left[\frac{1}{mm}\right]$	$f\left[\frac{mm^3}{N}\right]$
20	-190.1	5.984	1.726	-0.03303	0.02399	0.001433
35	-231.1	3.485	2.906	-0.00814	0.01683	0.0004
50	-255.9	3.879	2.891	-0.0096	0.01908	0.000625
60	-269.9	3.735	3.024	-0.00816	0.02133	0.00022

Dep

APPROXIMATING EQUATION SUGGESTION FOR SPRING STIFFNESS FORMULA $K = -200 \left[\frac{N}{mm} \right] + 4 \left[\frac{N}{mm^3} \right] A + 2.5 \text{ [mm]} E_C + 0.009(-A^2 \left[\frac{N}{mm^5} \right] + A.E_C \left[\frac{1}{mm} \right] + E_C^2 \left[\frac{mm^3}{N} \right])$

The error of the simplified formula in case of internal plate





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DESIGN RESISTANCE

Test-based design resistance EC 1993-1-3 2006 (A.6.3.2 (1))

Buckling mode of C90-0.9-600-A



Design values for $\alpha = b/66$

Comparison of local failure at the end of the braced specimen between test and FEA

Cranative and trues	Test-based design	Numerical design resistance [KN]			
Specimens type	resistance [KN]	L=300 [mm]	L=600 [mm]	L=2000 [mm]	
C90-0.9-A	26.50	27.22 (2.7%)	27.23 (2.7%)	27.16 (2.5%)	
C90-0.9-X	26.50	27.17 (2.5%)	27.38 (3.3%)	27.46 (3.6%)	
C140-0.9-WM	27.33	27.13 (0.7%)	27.37 (0.1%)	27.34 (0.0%)	
C140-1.5-WM	63.24	59.40 (6.0%)	59.97 (5.1%)	59.87 (5.3%)	







FAILURE MODES COMPARISON

Contact

pair

Comparison of connection failure of the unbraced specimen between test and FEA

Comparison of connection failure of the braced specimen between test and FEA



CONCLUSIONS

- Simplified numerical models were developed for fully and partially-PAC filled CFS columns and joints.
- A simplified numerical formula was proposed to define the solid-replacement spring stiffness.
- Design equivalent geometrical imperfection amplitudes were suggested to predict the design resistance of PAC filled thin-walled columns and joints.
- Further studies

- **PUBLICATIONS** [1] ANALYSIS AND CROSS SECTION DEVELOPMENT OF COLD-FORMED **STEEL RECTANGULAR HOLLOW FLANGE BEAMS, 15th International** Miklos Ivanyi PhD & DLA Symposium, Pecs, Hungary. [2] ANALYSIS AND CROSS SECTION DEVELOPMENT OF COLD-FORMED **STEEL RECTANGULAR HOLLOW FLANGE BEAMS, Nathalie Eid, Attila** László Joó, Pollack Periodica journal, published. [3] NUMERICAL SIMULATION OF ULTRA-LIGHTWEIGHT-CONCRETE **ENCASED COLD-FORMED STEEL STRUCTURES**, Eurosteel 2021, Sheffield, UK, published. [4] Simplified model of ultra-lightweight-concrete encased coldformed steel structures Nathalie Eid, Attila László Joó, Advances in
- **Civil Engineering**, published.

Thank you for your listening!



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